

THE FOUR POINT SHEAR TEST ON SINGLE NOTCHED SPECIMENS:  
AN EXPERIMENTAL AND NUMERICAL ANALYSIS.

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An experimental and numerical investigation is carried out for a particular testing geometry: the single edge notched specimen subjected to four point shear, where a Mixed Mode crack propagation is activated in concrete, if the specimen is not too small and the aggregates are not too large. When both such conditions are verified, each crack growth step is governed locally by a Mode I strain field and the energy dissipation is tendentially provided by the fracture energy  $G_f$ , whereas permanent deformation in the bulk material and interlocking on the crack surface, result to be negligible.

INTRODUCTION

The single edge notched specimen subjected to four point shear, firstly proposed in (1) for concrete, was then repropoed in (2-5) with two symmetrical initial cracks—the so-called double edge notched specimen subjected to four point shear. Purpose of these studies was analyzing the different failure mechanisms involved by varying geometrical ratios and size-scale of the specimen. The Mixed Mode crack propagation is thus in competition with other two failure mechanisms:

- (1) the bending failure at the supports;
- (2) the splitting failure at the specimen center.

With large-sizes, Mixed Mode crack propagation is favoured and prevails over the two strength overcoming mechanisms. Even with small aggregates, Mixed Mode crack propagation prevails and the energy dissipation is confined to the crack surface. In both cases, Mixed Mode

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fracture energy and Mode I fracture energy  $G_F$  result to be very close, since the additional energy dissipation by permanent deformation in the bulk material, as well as by friction and interlocking, tends to be negligible.

#### MATERIAL PROPERTIES AND EXPERIMENTAL PROGRAM

Two different concretes, with maximum aggregate size  $D_{max} = 10$  and 20mm respectively, were selected to extend the experimental program published in (6).

Three point bending tests were carried out to obtain the fracture energy  $G_F$  according to the RILEM Recommendation (7). The same fracture toughness parameter was determined also by stable direct tension tests on prenotched cylindrical specimens of diameter  $\phi = 10$ cm and height  $H = 20$ cm. The related experimental results, averaged over four identical specimens, are reported in Table 1.

TABLE 1: Mode I Fracture energy  $G_F$

Testing procedure	Specimen sizes	Concrete	Fracture energy $G_F$ (N/m)
Stable three-point bending test	10·10·84 cm	1	116.7 ± 12.4
		2	117.8 ± 19.9
Stable tension test	$\phi - 10$ H - 20	1	125.7 ± 7.3
		2	124.2

Four identical single edge notched shear specimens (Fig.1) were tested for each case of Table 2. The specimens were kept in a controlled environment at 20°C and 95% relative humidity up to four hours before testing. The notches were performed by means of a circular saw. Each specimen was provided with four steel supports of sizes 2x2x10 cm, glued at the four intended loading points, and with displacement transducers (LVDT) to measure the following quantities (6).

- Two transducers to measure the deflections  $\delta_1$  and  $\delta_2$  of the two upper loading points.
- Two transducers (one per each side of the beam) to measure the crack mouth sliding displacement (CMSD).

- One central transducer to measure the crack mouth opening displacement (CMOD).

TABLE 2: Testing geometries

Concrete	D <sub>max</sub> (mm)	Specimen	Thickness t (cm)	Depth b (cm)	Length L (cm)	Span l (cm)	Crack length a (cm)	Support distance c (cm)
1	10	1 B	10	20	84	80	4	8
		1 C	10	30	125	120	6	12
2	20	2 C	10	30	125	120	6	12

The tests were carried out using a servocontrolled machine (max.load=60t) and the deflection  $\delta_2$  as feedback signal (6). The imposed deflection rate was approximately 0.025  $\mu\text{m}/\text{sec}$ .

The following diagrams were recorded.

- P vs  $\delta_1$  and P vs  $\delta_2$ , P being the total load, from which it is possible to obtain the curves  $F_1$  vs  $\delta_1$  and  $F_2$  vs  $\delta_2$ , for the evaluation of the Mixed Mode fracture energy (Fig.2).
- P vs CMSD, where CMSD is the average between the two transducers on the two beam sides, in order to take into account torsional effects.
- P vs CMOD.

The results are displayed in Table 3, for concrete 1 and 2 respectively. Mixed Mode fracture energy is defined as the ratio of total dissipated energy to total fracture area. Mixed Mode fracture energy results to be higher than Mode I fracture energy, ~19% with small aggregates (specimen 1C) and ~33% with large aggregates (specimen 2C), the specimen depth being equal to 30 cm. This is very likely due to additional energy dissipation by interlocking, which is higher with larger aggregates. On the other hand, the increase is ~19% with large specimens (1C) and ~31% with small specimens (1B), for concrete 1. The interlocking effect appears to increase with decreasing specimen size.

The same trends were found in the preceding experimental investigation (6).

#### NUMERICAL SIMULATION

The experimental tests were simulated by a finite element procedure, where the controlling parameter is the crack length (8-10). For mixed mode non-collinear crack propagation a continuous modification of the mesh is required. A finite element rosette is roto-translated at each crack growth step. Fig.3 shows three different steps of the crack propagation process in specimen 2C. The presence of cohesive strain-softening forces is indicated by the connections. At the first step the cohesive zone is missing and the load  $P$  producing the ultimate tensile stress at the crack tip is computed. Such a value,  $P$ , together with the related loading point displacement,  $\delta$ , gives the first point of the  $P$  vs  $\delta$  diagram. The fictitious crack tip propagates by a pre-defined length  $\Delta a$  in a direction orthogonal to the maximum circumferential stress (11). Then, the load  $P$  producing the ultimate tensile stress at the up-dated fictitious crack tip and the related deflection  $\delta$  are computed at each step in order to obtain the subsequent points of the diagram.

The substructures used in the numerical simulation are shown in Fig.4, whereas the deformed configurations at the computation steps 6, 14 and 26, are represented in Fig.5, with a magnification factor equal to 300. The material properties assumed in the numerical analysis are the following:

Young's modulus,  $E = 40,000$  MPa  
 Poisson ratio,  $\nu = 0.1$   
 Ultimate tensile stress,  $\sigma_U = 2$  MPa  
 Fracture energy,  $G_F = 125$  N/m

It is worthwhile to remark that the propagation of the real crack tip begins only at the step 29, when the uncracked ligament is less than 1/5 of the beam depth.

The numerical load vs deflection diagrams,  $F_1$  vs  $\delta_1$  and  $F_2$  vs  $\delta_2$ , are reported in Fig.2, together with the experimental ones, averaged over four specimens. The loading point 1 (Fig.1) undergoes a quasi-elastic unloading with small permanent deformation, while the loading point 2 undergoes a softening unloading with ever increasing deformation.

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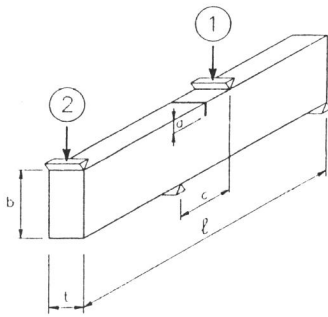


Figure 1 Single edge notched specimen subjected to four point shear.

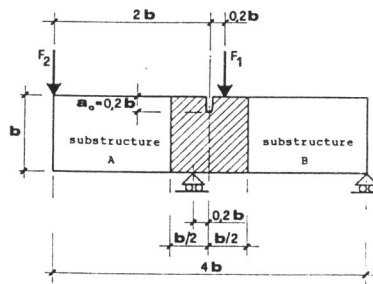


Figure 4 Substructures used in the numerical simulation.

Table 3 - Comparison between mixed-mode fracture energy and mode I fracture energy.

D max (mm)	Specimen	Maximum load P (KN)	Dissipated energy $W_0$ (N.m)	Fracture area S (m <sup>2</sup> )	Mixed-mode fracture energy $G_{F^*}$ (N/m)	Mode I fracture energy $G_F$ (N/m)	$\frac{G_{F^*} - G_F}{G_F} \cdot 100$ (%)
10	1 B	56.16	2.182	0.01939	112.53	125.7	31
		69.16	3.899	0.01967	198.22		
		59.11	3.287	0.01958	164.3		
		68.18	2.826	0.02001	167.88 ± 51.2		
	1 C	85.35	4.769	0.02965	160.84	125.7	19
		82.89	4.727	0.03003	157.41		
		83.39	4.512	0.02893	155.96		
		79.95	3.568	0.02835	125.86		
20	2 C	98.10	5.390	0.02830	190.46	124.2	33
		83.39	4.048	0.02847	142.18		
	93.69	5.651	0.03040	185.89	165.2 ± 26.6		
	80.93	4.169	0.02929	142.34			

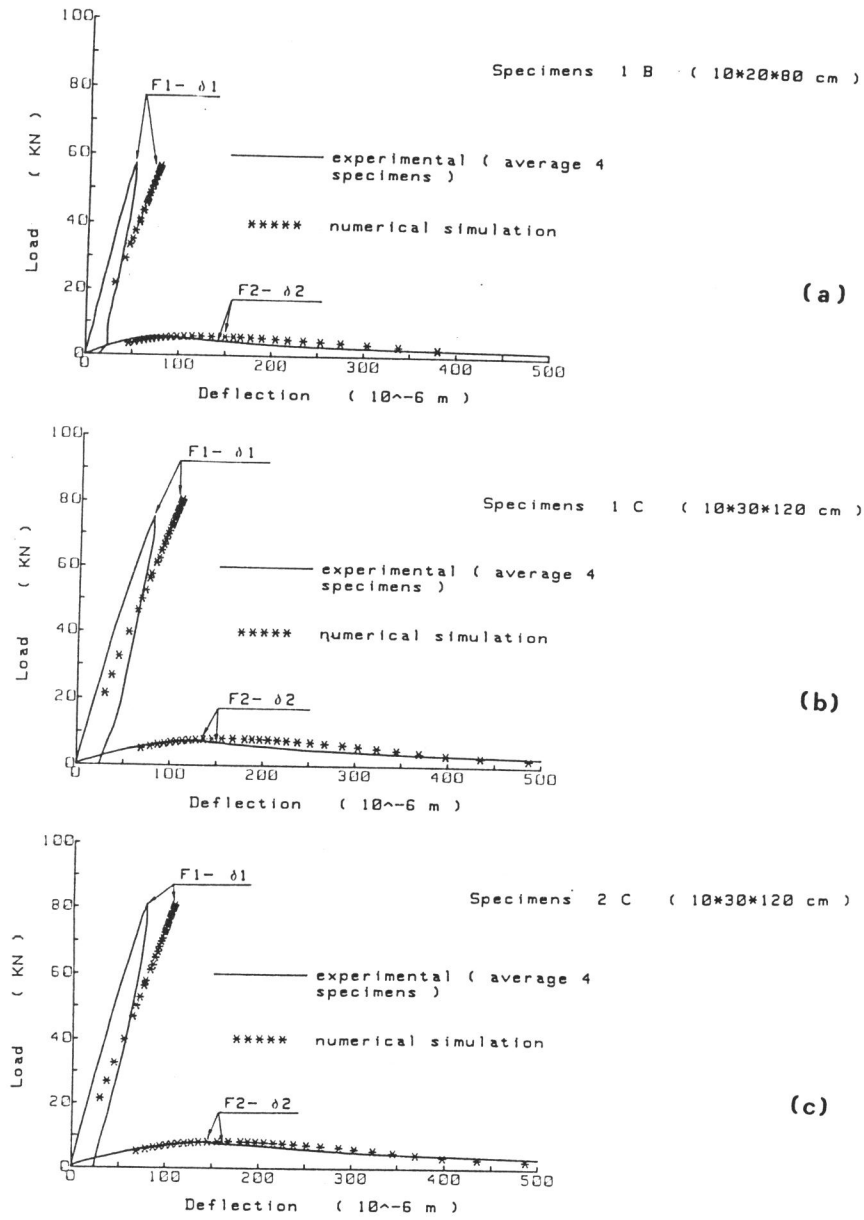


Figure 2 Experimental and numerical load vs deflection diagrams.



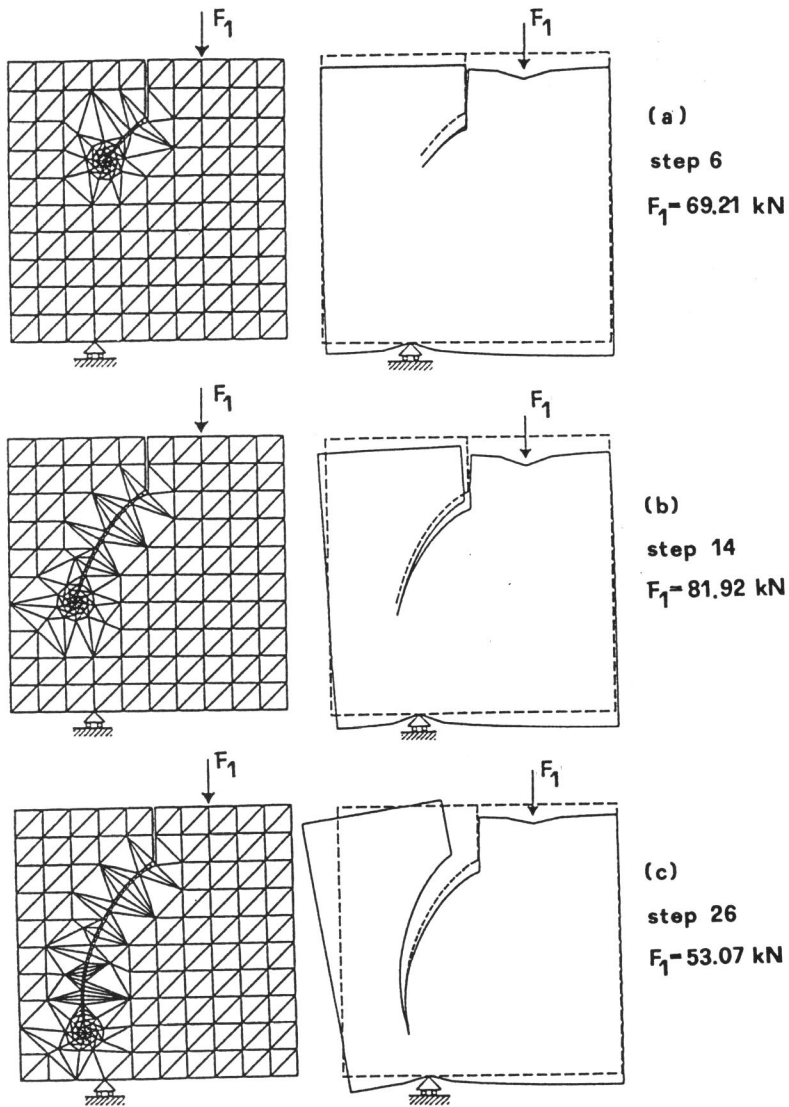


Figure 3 Finite element remeshing (specimen 2c)

Figure 5 Subsequent deformed configurations (specimen 2c)